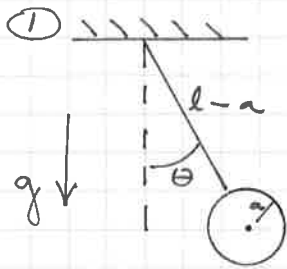


1983

Ph 205 SOLUTIONS TO FINAL EXAM

✓



THE PENDULUM IS A SIMPLE PENDULUM NOT A DOUBLE....
THE KEY INSIGHT IS THAT THE ICE ROTATES AS THE PENDULUM SWINGS, WHILE THE WATER DOES NOT. IN THE CASE OF WATER THERE IS NO FRICTION \Rightarrow NO TORQUE ON WATER \Rightarrow NO ROTATION!

$$T_{\text{WATER}} = \frac{1}{2} m v^2 = \frac{1}{2} m l^2 \dot{\theta}^2$$

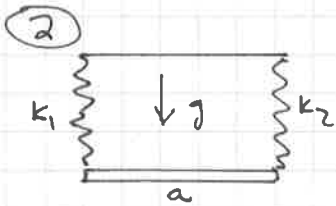
$$T_{\text{ICE}} = \frac{1}{2} m v^2 + \frac{1}{2} I \dot{\theta}^2 = \frac{1}{2} m l^2 \dot{\theta}^2 + \frac{1}{2} \frac{2}{5} m a^2 \dot{\theta}^2 \quad \left. \vphantom{T_{\text{ICE}}} \right\} \equiv \frac{K}{2} m l^2 \dot{\theta}^2$$

$$V = m g l (1 - \cos \theta)$$

$$\text{so } K m l^2 \ddot{\theta} = -m g l \sin \theta$$

FOR SMALL θ WE HAVE SIMPLE HARMONIC MOTION WITH $\omega = \sqrt{\frac{g}{K l}}$

$$T = \frac{2\pi}{\omega} \quad \text{so} \quad \frac{T_{\text{ICE}}}{T_{\text{WATER}}} = \sqrt{\frac{K_{\text{ICE}}}{K_{\text{WATER}}}} = \sqrt{1 + \frac{2}{5} \frac{a^2}{l^2}}$$



UNDER THE RESTRICTIONS STATED THERE ARE ONLY 2 DEGREES OF FREEDOM \Rightarrow 2 MODES

a) $k_1 = k_2$. WE USE ELEMENTARY MEANS, TO GET A CHECK ON LAGRANGE'S METHOD IN b)

MODE 1: $x_1 \downarrow \uparrow x_2$ $x_1 = x_2 \equiv x$ NO ROTATION

$$\text{C.M. MOTION: } m \ddot{x} = -(k_1 + k_2) x = -2Kx$$

$$\omega = \sqrt{\frac{2K}{m}}$$

MODE 2: $x_1 = -x_2$ CM AT REST; ROTATION



$$I \ddot{\theta} = N_{\text{ABOUT CM}} = -2K x_1 \left(\frac{a}{2}\right) = -\frac{1}{2} K a^2 \theta \quad \text{WITH } \theta = \frac{2x_1}{a}$$

$$I_{\text{CM}} = \frac{1}{12} m a^2 \quad \text{so} \quad \ddot{\theta} = -\frac{6K}{m} \theta \quad \Rightarrow \quad \omega = \sqrt{\frac{6K}{m}}$$

b) $k_1 \neq k_2$. AT EQUILIBRIUM WE MUST HAVE $F_1 = F_2$ SO NO TORQUE AND $F_1 + F_2 = Mg$

$$\text{so } k_1 x_1 = k_2 x_2 \quad \text{AND} \quad k_1 x_1 + k_2 x_2 = Mg \quad \Rightarrow \quad x_1 = \frac{Mg}{2k_1} \quad x_2 = \frac{Mg}{2k_2}$$

WHERE x_1 AND x_2 ARE MEASURED FROM THE ENDS OF THE UNSTRETCHED SPRINGS.

IF WE SET $q_1 = x_1 - \frac{Mg}{2k_1}$ AND $q_2 = x_2 - \frac{Mg}{2k_2}$

THEN $V = \frac{1}{2} k_1 q_1^2 + \frac{1}{2} k_2 q_2^2$

AND $T = T_{OF CM} + T_{ROTATION} = \frac{1}{2} M \left(\frac{\dot{q}_1 + \dot{q}_2}{2} \right)^2 + \frac{1}{2} \cdot \frac{1}{12} M a^2 \left(\frac{\dot{q}_1 - \dot{q}_2}{a} \right)^2$
↑ NOT 2a
 $= \frac{1}{2} M \left\{ \frac{1}{3} \dot{q}_1^2 + \frac{1}{3} \dot{q}_1 \dot{q}_2 + \frac{1}{3} \dot{q}_2^2 \right\}$

SO THE EQUATIONS OF MOTION ARE $\frac{M}{3} \ddot{q}_1 + \frac{M}{6} \ddot{q}_2 = -k_1 q_1$

$\frac{M}{6} \ddot{q}_1 + \frac{M}{3} \ddot{q}_2 = -k_2 q_2$

TRY $q_1 = A \cos \omega t$ $q_2 = B \cos \omega t$

$\left(\frac{k_1}{M} - \frac{1}{3} \omega^2 \right) A - \frac{\omega^2}{6} B = 0$

$-\frac{\omega^2}{6} A + \left(\frac{k_2}{M} - \frac{1}{3} \omega^2 \right) B = 0$

SET DETERMINANT TO ZERO $\Rightarrow \frac{k_1 k_2}{M^2} - \frac{1}{3} \left(\frac{k_1 + k_2}{M} \right) \omega^2 + \frac{1}{12} \omega^4 = 0$

$\omega^4 - 4 \left(\frac{k_1 + k_2}{M} \right) \omega^2 + 12 \frac{k_1 k_2}{M^2} = 0$

$\omega^2 = +2 \left(\frac{k_1 + k_2}{M} \right) \pm \frac{1}{M} \sqrt{4(k_1 + k_2)^2 - 12 k_1 k_2}$

CHECK $k_1 = k_2 = k \Rightarrow \omega^2 = \frac{+4k \pm 2k}{M} = \frac{2k}{M}, \frac{6k}{M}$ ✓

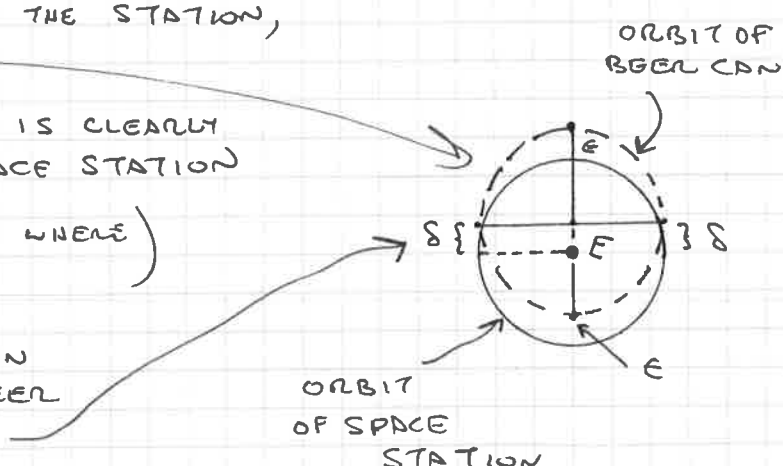
3) BEER CAN IN ORBIT WE CAN SOLVE THIS BY RELATIVELY ELEMENTARY CONSIDERATIONS.

SO LONG AS THE BEER CAN IS NOT THROWN EXTREMELY FAST, IT WILL ORBIT THE EARTH IN AN ELLIPSE (WE IGNORE THE SUN). IF THIS ELLIPSE IS TO APPEAR TO "ORBIT" THE STATION, IT MUST LOOK LIKE

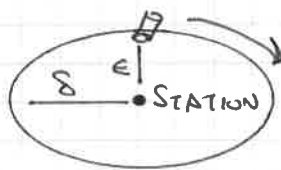
THE PERIOD OF THE ORBIT IS CLEARLY THE SAME AS THAT OF THE SPACE STATION

(KEPLER'S 3RD LAW $T^2 \sim a^2$ WHERE)
 $a =$ SEMI MAJOR AXIS

NOTE THAT WHEN THE STATION WAS MOVED THRU 90° , THE BEER CAN IS BEHIND THE STATION BY AMOUNT δ .



RELATIVE TO THE STATION THE BEER CAN FOLLOWS A PATH



MOTION OF STATION REL. TO EARTH

IF $\epsilon = \delta$ THE PATH IS A CIRCLE.

TO EARTH

\therefore THE ASTRONAUT SHOULD THROW THE BEER CAN PARALLEL TO THE ORBIT OF THE STATION ABOUT THE EARTH, AND OPPOSITE TO THE MOTION OF THE STATION.

TO FIND THE VELOCITY, WE NOTE THAT L_{CAN} ABOUT EARTH \approx SAME AS ANGULAR MOMENTUM OF A CAN ON THE STATION. (WHEN THE STATION HAS MOVED BY 90° $r_{CAN} = r_0$, $v_{\odot CAN} = v_{\odot STATION}$)

$\Rightarrow L_{CAN} = m r_0^2 \omega_0$ WHERE $\omega_0^2 = \frac{GM_E}{r_0^3}$

SO $m(r_0 + \epsilon) \omega_i = m r_0^2 \omega_0$

$(r_0^2 + 2\epsilon r_0) \omega_i = r_0^2 \omega_0$

$\omega_i \approx \omega_0 (1 - \frac{2\epsilon}{r_0})$

$v_i = (r_0 + \epsilon) \omega_i = r_0 \omega_0 (1 + \frac{\epsilon}{r_0}) (1 - \frac{2\epsilon}{r_0}) \approx v_0 (1 - \frac{\epsilon}{r_0})$

$\therefore \Delta v = -\epsilon \frac{v_0}{r_0} = -\epsilon \omega_0 = -\epsilon \sqrt{\frac{GM_E}{r_0^3}}$

IF YOU CLAIM THE ASTRONAUT WAS VELOCITY $= (r_0 + \epsilon) \omega_0 = v_0 (1 + \frac{\epsilon}{r_0})$ THEN $\Delta v = -2\epsilon \frac{v_0}{r_0} = -2\epsilon \omega_0$

WE ASSUME THE ASTRONAUT HAD THE SAME VELOCITY AS STATION - OTHERWISE HE DRIFTS AWAY...

FINALLY WE NEED δ . THE ELLIPSE OF THE BEER CAN IS

$r \approx r_0 + \epsilon \cos \omega_0 t$ (REGULAR FORM OF SOLUTION TO SMALL OSCILLATION ABOUT CIRCULAR ORBIT)

AND CONSERVATION OF ANGULAR MOMENTUM \Rightarrow

$\dot{\theta} = \frac{L}{m r^2} = \frac{L}{m r_0^2 (1 + \frac{\epsilon}{r_0} \cos \omega_0 t)^2} \approx \omega_0 (1 - \frac{2\epsilon}{r_0} \cos \omega_0 t)$

$\Rightarrow \theta = \omega_0 t + \frac{2\epsilon \omega_0}{r_0} \sin \omega_0 t$

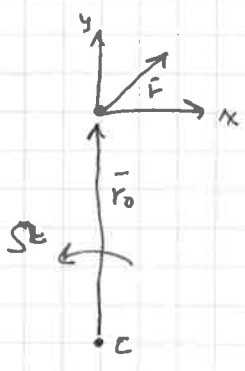
$\Delta \theta = \theta - \omega_0 t = -2 \frac{\epsilon}{r_0} \sin \omega_0 t$

δ IS MEASURED WHEN $\omega_0 t = 90^\circ \Rightarrow \Delta \theta = -\frac{2\epsilon}{r_0}$

AND $\underline{\underline{\delta}} = r_0 \Delta \theta = -2\epsilon$

THE 'ORBIT' IS AN ELLIPSE NOT A CIRCLE

SOLUTION IN THE ROTATING FRAME



IN THE ROTATING FRAME

$$m \bar{a} = \bar{F} - m \bar{\Omega} \times \bar{\Omega} \times (\bar{r}_0 + \bar{r}) - 2m \bar{\Omega} \times \bar{v}$$

FOR $\bar{r} \ll \bar{r}_0$ THIS IS APPROXIMATELY

$$m \bar{a} = \frac{-GMm}{(r_0+y)^2} \hat{y} + m \Omega^2 (r_0+y) \hat{y} - 2m \Omega \hat{z} \times (\dot{x} \hat{x} + \dot{y} \hat{y})$$

$$= \left[-\frac{GMm}{r_0^2} \left(1 - \frac{2y}{r_0}\right) + m \Omega^2 (r_0+y) \right] \hat{y} + 2m \Omega (\dot{y} \hat{x} - \dot{x} \hat{y})$$

NOW $m \Omega^2 r_0 = \frac{GMm}{r_0^2}$

IF INCLUDE \hat{z} DEPENDENCE OF GRAVITY

SO $m \bar{a} = 3m \Omega^2 y \hat{y} + 2m \Omega (\dot{y} \hat{x} - \dot{x} \hat{y}) - m \Omega^2 z \hat{z}$

WE TRY AN ELLIPSE-LIKE SOLUTION WHICH MIGHT BE APPROPRIATE TO THE INITIAL CONDITION $x(0) = 0, \dot{y}(0) = \epsilon$

$x = A \sin \omega t \quad y = \epsilon \cos \omega t$

x: $\ddot{x} = 2 \Omega \dot{y} \Rightarrow -A \omega^2 = -2 \Omega \omega \epsilon \Rightarrow A = \frac{2 \Omega}{\omega} \epsilon$

y: $\ddot{y} = 3 \Omega^2 y - 2 \Omega \dot{x} \Rightarrow -\epsilon \omega^2 = 3 \Omega^2 \epsilon - 2 \Omega \omega A$

$$= 3 \Omega^2 \epsilon - 4 \Omega^2 \epsilon = -\Omega^2 \epsilon$$

$\therefore \omega = \Omega \quad \text{AND} \quad A = 2 \epsilon$

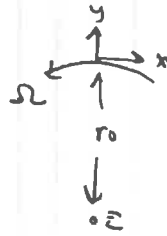
(OR $\omega = -\Omega$ AND $A = -2 \epsilon$ WHICH COMES TO THE SAME THING)

THE INITIAL VELOCITY IS $\dot{x}(0) = 2 \epsilon \Omega \quad \dot{y}(0) = 0$

$$= 2 \epsilon \sqrt{\frac{GM}{r_0^3}}$$

NOTE THAT $\left. \begin{matrix} \dot{x}(0) > 0 \\ \dot{y}(0) = 0 \end{matrix} \right\} \Rightarrow$ MOTION OPPOSITE TO THAT OF THE STATION, BUT PARALLEL TO THE STATION'S ORBIT

GARBAGE IS THROWN OUT OF A SPACE STATION IN A CIRCULAR ORBIT OF RADIUS r_0 ABOUT THE EARTH. DESCRIBE THE MOTION OF THE GARBAGE SUPPOSING IT IS GIVEN SMALL INITIAL VELOCITY V_x, V_y OR V_z RELATIVE TO THE SPACE STATION



QUICK SOLUTIONS!

a) $\Delta V = V_z$. $L_z' = \text{SAME AS } L_z$ FOR MOTION IN A PLANE TILTED BY

$$\theta = \frac{V_z}{\Omega r_0}, \quad E \text{ REMAINS SAME TO 1ST ORDER IN } V_z$$

(KE \rightarrow KE + $\frac{1}{2} m V_z^2$; PE = SAME)

\therefore ORBIT REMAINS A CIRCLE, BUT TILTED BY θ

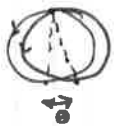


b) $\Delta V = V_y$. ORBIT IS AN ELLIPSE IN THE ORIGINAL PLANE AGAIN KE \sim SAME, PE \neq SAME $\Rightarrow E = \text{SAME}$ ALSO $L = \text{SAME}$

\therefore MAJOR AXIS OF ELLIPSE REMAINS $2r_0$

FOR SMALL ECCENTRICITY, ELLIPSE \sim CIRCLE

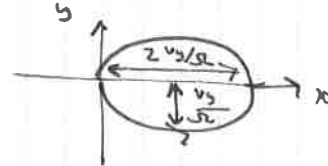
\therefore NEW ORBIT SAME AS ORIGINAL BUT ROTATED BY $\theta = \frac{V_y}{\Omega r_0}$



$$\Delta \text{ PERIGEE} = r_0 \theta = \frac{V_y}{\Omega}$$

FOR $V_y < 0$

THE MOTION RELATIVE TO THE STATION, AS MEASURED IN THE ROTATING $x-y$ FRAME IS (FOR $V_y < 0$)



c) $\Delta V = V_x < 0$. NOW KE $>$ KE $_i$, $L_z' >$ L_z \Rightarrow ORBIT IS BIGGER & HAS LONGER PERIOD.

$$\frac{PE \approx \text{SAME}}{E > E_i}$$

FOR $V_x < 0$ THE ORIGINAL PT IS THE PERIGEE

(FOR $V_x > 0$, IT WOULD BE APOGEE.)

FOR SMALL ECCENTRICITY THE ORBIT IS STILL A CIRCLE.

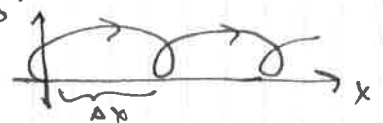
$$\text{NOW } m \left(\frac{V_i + V_x}{R} \right)^2 = \frac{GM}{r_0^2} = \frac{V_i^2}{r_0} \Rightarrow R \approx r_0 \left(1 + \frac{2V_x}{V_i} \right)$$



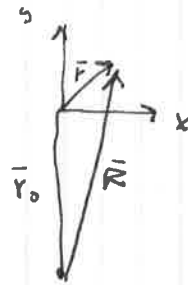
$$\text{APOGEE IS } 2R - r_0 = r_0 \left(1 + 4 \frac{V_x}{V_i} \right) \quad V_i = r_0 \Omega$$

IN THE ROTATING FRAME ORBIT LOOKS LIKE

$$\Delta x \approx V_i \Delta T = V_i T_i \left[\left(\frac{R}{r_0} \right)^2 - 1 \right] = V_i T_i \cdot 3 \frac{V_x}{V_i} = 3V_x T_i = \frac{6\pi V_i}{\Omega}$$



SOLUTION IN THE ROTATING FRAME



$$\bar{a} = \frac{\bar{F}}{m} - \bar{\Omega} \times (\bar{\Omega} \times \bar{r}) - 2\bar{\Omega} \times \bar{v}$$

$$\mathbf{R} = x \hat{x} + (r_0 + y) \hat{y} + z \hat{z}$$

$$\bar{\Omega} = \Omega \hat{z} \quad \bar{v} = \dot{x} \hat{x} + \dot{y} \hat{y}$$

$$\frac{\bar{F}}{m} = -\frac{GM}{R^3} \bar{R} = -\frac{GM}{R^2} \left(\frac{x}{R} \hat{x} + \frac{r_0 + y}{R} \hat{y} + \frac{z}{R} \hat{z} \right)$$

$$\approx -\frac{GM}{r_0^3} x \hat{x} \Rightarrow \frac{GM y}{(r_0 + y)^2} \Rightarrow \frac{GM z}{r_0^3} \hat{z}$$

$$\frac{GM}{r_0^3} \approx \Omega^2$$

$$\approx -\Omega^2 x \hat{x} + \Omega^2 r_0 \left(1 - \frac{2y}{r_0}\right) \hat{y} + \Omega^2 z \hat{z}$$

$$\bar{\Omega} \times \bar{R} = -\Omega(r_0 + y) \hat{x} + \Omega x \hat{y} \quad \bar{\Omega} \times (\bar{\Omega} \times \bar{R}) = -\Omega^2(r_0 + y) \hat{y} - \Omega^2 x \hat{x}$$

$$\bar{\Omega} \times \bar{v} = \Omega \dot{x} \hat{y} - \Omega \dot{y} \hat{x}$$

$$\text{so } \bar{a} = -\Omega^2 x \hat{x} - \Omega^2 r_0 \hat{y} + 2\Omega^2 y \hat{y} - \Omega^2 z \hat{z} + \Omega^2 (r_0 + y) \hat{y} + \Omega^2 x \hat{x} + 2\Omega \dot{y} \hat{x} + 2\Omega \dot{x} \hat{y}$$

$$\bar{a} = 2\Omega \dot{y} \hat{x} + (3\Omega^2 y - 2\Omega \dot{x}) \hat{y} - \Omega^2 z \hat{z}$$

$$\ddot{x} = 2\Omega \dot{y}$$

$$\ddot{y} = 3\Omega^2 y - 2\Omega \dot{x}$$

$$\ddot{z} = -\Omega^2 z$$

INITIAL CONDITIONS: $\bar{r} = 0 \quad \dot{\bar{r}} = (v_{0x}, v_{0y}, v_z)$

AT ONCE $z = \frac{v_z}{\Omega} \sin \Omega t$

FOR x & y WE TRY OSCILLATORY SOLUTIONS - WITH A POSSIBLE OFFSET.

IN ADDITION, THE SOLUTIONS IN x MAY DRIFT ...

$$x = A + B \cos \omega t + C \sin \omega t + Dt$$

$$y = E + F \cos \omega t + G \sin \omega t$$

$$x(0) = 0 = A + B$$

$$y(0) = 0 = E + F$$

$$\dot{x} = A(1 - \omega \sin \omega t) + C \cos \omega t + D$$

$$\dot{y} = E(1 - \omega \sin \omega t) + G \cos \omega t$$

$$\ddot{x} = 2\Omega \dot{y} \Rightarrow A\omega^2 \cos \omega t - C\omega^2 \sin \omega t = 2E\Omega \sin \omega t + 2G\Omega \cos \omega t$$

so $A\omega = 2G\Omega$ AND $C\omega = -2E\Omega$

$$\ddot{y} = 3\Omega^2 y - 2\Omega \dot{x} \Rightarrow E\omega^2 \cos \omega t - G\omega^2 \sin \omega t = 3\Omega^2 E(1 - \cos \omega t) + 3\Omega^2 G \sin \omega t - 2\Omega \omega A \sin \omega t - 2\Omega \omega C \cos \omega t \Rightarrow 2D\Omega$$

$$s. \quad E\omega^2 = -3E\Omega^2 + 4E\Omega^2 = E\Omega^2 \Rightarrow \omega = \Omega$$

$$-G\omega^2 = 3G\Omega^2 - 4G\Omega^2 \Rightarrow \omega = \Omega$$

$$AND \quad 3\Omega^2 E - 2D\Omega = 0 \Rightarrow D = \frac{3}{2} E\Omega$$

$$x = 2G(1 - \cos \Omega t) - 2E \sin \Omega t + \frac{3}{2} E\Omega t$$

$$y = E(1 - \cos \Omega t) + G \sin \Omega t$$

$$\dot{x}(0) = -2E\Omega + \frac{3}{2} E\Omega = -\frac{1}{2} E\Omega = v_x \Rightarrow E = \frac{-2v_x}{\Omega}$$

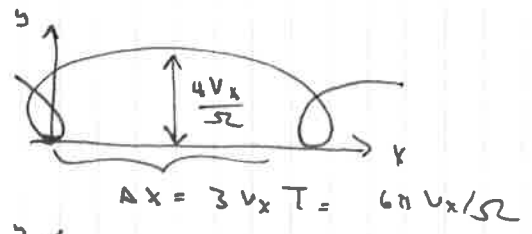
$$\dot{y}(0) = G\Omega = v_y \Rightarrow G = \frac{v_y}{\Omega}$$

$$x = \frac{2v_y}{\Omega} (1 - \cos \Omega t) + 4 \frac{v_x}{\Omega} \sin \Omega t - 3v_x t$$

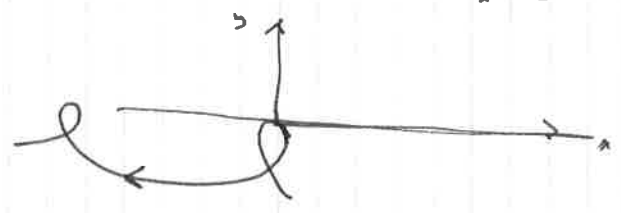
$$y = -\frac{2v_y}{\Omega} (1 - \cos \Omega t) + \frac{v_y}{\Omega} \sin \Omega t$$

$$z = \frac{v_z}{\Omega} \sin \Omega t$$

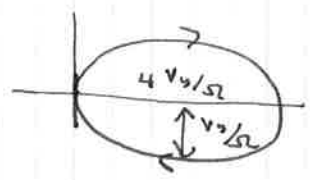
$v_x < 0, v_y = 0$



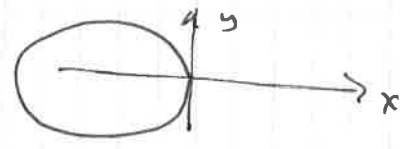
$v_x > 0, v_y = 0$



$v_x = 0, v_y > 0$

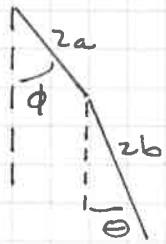


$v_x = 0, v_y < 0$



← SPACE STATION
MOVES IN THE
-X DIRECTION.

④ a) WE USE LAGRANGE TO FIND THE EQUATION OF MOTION OF θ .



THIS IS A BIT OF A GRIND:

$$V = m g a (1 - \cos \phi) + m g [2a(1 - \cos \phi) + b(1 - \cos \theta)] + \text{const}$$

$$= m g b (1 - \cos \theta) + f(\phi)$$

$$T_a = \frac{1}{2} m a^2 \dot{\phi}^2 + \frac{1}{2} \cdot \frac{1}{12} (2a)^2 \dot{\phi}^2 = \frac{1}{2} \cdot \frac{4}{3} m a^2 \dot{\phi}^2$$

$$T_b = \frac{1}{2} m v_{cm}^2 + \frac{1}{2} \cdot \frac{1}{3} m b^2 \dot{\theta}^2$$

$$x_b = 2a \sin \phi + b \sin \theta$$

$$\dot{x}_b = 2a \cos \phi \dot{\phi} + b \cos \theta \dot{\theta}$$

$$y_b = 2a \cos \phi + b \cos \theta$$

$$\dot{y}_b = -2a \sin \phi \dot{\phi} - b \sin \theta \dot{\theta}$$

$$v_b^2 = 4a^2 \dot{\phi}^2 + 4ab \cos(\theta - \phi) \dot{\phi} \dot{\theta} + b^2 \dot{\theta}^2$$

$$\text{so } L = \frac{1}{2} \frac{4}{3} m b^2 \dot{\theta}^2 + 2mab \cos(\theta - \phi) \dot{\theta} \dot{\phi} - m g b (1 - \cos \theta) + F(\phi)$$

$$\frac{\partial L}{\partial \dot{\theta}} = \frac{4}{3} m b^2 \dot{\theta} + 2mab \cos(\theta - \phi) \dot{\phi}$$

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{\theta}} = \frac{4}{3} m b^2 \ddot{\theta} + 2mab (\cos(\theta - \phi) \ddot{\phi} - \sin(\theta - \phi) (\dot{\theta} - \dot{\phi}) \dot{\phi})$$

$$= \frac{\partial L}{\partial \theta} = -m g b \sin \theta - 2mab \sin(\theta - \phi) \dot{\theta} \dot{\phi}$$

WHAT A MESS!

$$b) \quad t_{I \rightarrow II} = t_{III \rightarrow IV} = \frac{\pi}{\omega}$$

$$t_{II \rightarrow III} \sim t_{IV \rightarrow I} \sim \frac{\pi}{\Omega}$$

WHERE $\Omega =$ ANGULAR VELOCITY OF PENDULUM MOTION OF ROD 2b

$$\text{FROM ABOVE, WITH } \phi = \dot{\phi} = \ddot{\phi} = 0$$

$$\frac{4}{3} m b^2 \ddot{\theta} \sim -m g b \theta$$

$$\Rightarrow \Omega = \sqrt{\frac{3g}{4b}}$$

$$T_{I \rightarrow II} = \frac{2\pi}{\omega} + 2\pi \sqrt{\frac{4b}{3g}}$$

WHAT IS ω ?

THE KINETIC ENERGY OF ROD b AT TIME II MUST BE SUFFICIENT FOR THE ROD TO RISE TO THE VERTICAL

$$\Rightarrow KE \geq 2mgb$$

Ph 205 FINAL EXAM

IF $a = \frac{b}{2}$, ROD b IS ROTATING ABOUT ITS CM WHICH IS AT REST FROM t_I TO t_{II}

$$\therefore KE = \frac{1}{2} I_{cm} \omega^2 = \frac{1}{6} m b^2 \omega^2 \quad \text{JUST BEFORE } t_{II}$$

(IT'S CM WAS NOT FALLEN AT ALL \Rightarrow NO INCREASE IN K.E.)
DUE TO GRAVITY!

A QUICK USE OF THIS RESULT SUGGESTS SETTING

$$\frac{1}{6} m b^2 \omega^2 = 2 m g b \quad \Rightarrow \quad \omega^2 = \frac{12g}{b}$$

$$\text{AND SO } T_{I \rightarrow II} = \frac{2\pi}{\omega} \sqrt{\frac{b}{3g}} + 4\pi \sqrt{\frac{b}{3g}} = 5\pi \sqrt{\frac{b}{3g}} \approx 6 \text{ SEC}$$

FOR $b = 5 \text{ cm}$

HOWEVER, AT MOMENT t_{II} THE UPPER END OF ROD b BECOMES SUDDENLY FIXED \Rightarrow IMPULSE WHICH ERASES A LOT OF K.E.

DURING THE IMPULSE $L_{\text{ABOUT JOINT OF a TO b}} = \text{CONST}$

$$L_{\text{JOINT}} = L_{\text{OF CM MOTION}} + I_{cm} \omega = I_{cm} \omega = \frac{1}{3} m b^2 \omega$$

AFTER THE IMPULSE $L_{\text{JOINT}} = I_{\text{JOINT}} \omega' = \frac{4}{3} m b^2 \omega'$

$$\text{SO } \omega' = \frac{1}{4} \omega \quad \Rightarrow \quad KE_{\text{AFTER}} = \frac{1}{2} I_{\text{JOINT}} \omega'^2 = \frac{1}{2} \cdot \frac{4}{3} m b^2 \frac{\omega^2}{16}$$

$$= \frac{1}{24} m b^2 \omega^2 = 2 m g b$$

↑ TO RISE TO TOP

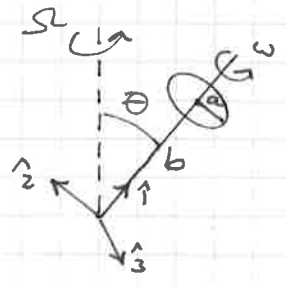
$$\text{SO } \omega^2 = 48g/b$$

$$\text{AND } T_{I-III} = \frac{2\pi}{\omega} \sqrt{\frac{b}{3g}} + 4\pi \sqrt{\frac{b}{3g}} = \frac{9\pi}{2} \sqrt{\frac{b}{3g}} \approx 5 \text{ SEC}$$

5

MOST PEOPLE PROCEEDED AS FOLLOWS:

SET UP A LOCAL COORD SYSTEM $\hat{1}, \hat{2}, \hat{3}$ WHICH ROTATES ABOUT THE VERTICAL AT RATE $\vec{\Omega} = (\Omega \cos \theta, \Omega \sin \theta, 0)$



CALCULATE TORQUES (AND MOMENTS OF INERTIA) ABOUT THE PIVOT POINT TO AVOID THE HORIZONTAL FORCE ON THE PIVOT (= $m\Omega^2 b \sin \theta$ FOR WHAT IT'S WORTH)

THEN $I_1 = ma^2$, $I_2 = I_3 = m(b^2 + \frac{a^2}{2})$ BY PARALLEL AXIS THM.

THE METHOD OF ANALYSIS IS A VARIATION ON EULER'S EQUATIONS:

$$\vec{N} = \frac{d\vec{L}}{dt} = \frac{\delta \vec{L}}{\delta t} + \vec{\Omega} \times \vec{L} \quad \text{WHERE } \frac{\delta \vec{L}}{\delta t} = \vec{I} \cdot \dot{\vec{\omega}}_{TOT} \quad \text{IN THE}$$

ROTATING FRAME. NOTE THAT WE USE $\vec{\Omega} \times \vec{L}$ NOT $\vec{\omega}_{TOT} \times \vec{L}$

$$\vec{N} = b \hat{1} \times (-mg \hat{z}) = -mgb \sin \theta \hat{3}$$

$$\text{SO } N_3 = I_3 \underset{0}{\dot{\omega}_3} + \Omega_1 I_2 \omega_2 - \Omega_2 I_1 \omega_1$$

THERE IS AN AMBIGUITY OF THE WORDING OF THE PROBLEM AS TO HOW TO DESCRIBE $\vec{\omega}_{TOT}$. I WAS INTENDED

$$\vec{\omega}_{TOT} = (\omega + \Omega \cos \theta, \Omega \sin \theta, 0) \quad \text{BUT } \vec{\omega}_{TOT} = (\omega, \Omega \sin \theta, 0)$$

WILL BE ACCEPTED ALSO. HERE I USE THE FIRST DEFINITION

$$\text{SO } -mgb \sin \theta = \Omega \cos \theta I_2 \Omega \sin \theta - \Omega \sin \theta I_1 (\omega + \Omega \cos \theta)$$

$$(I_2 - I_1) \Omega^2 \cos \theta \sin \theta - I_1 \omega \Omega \sin \theta + mgb \sin \theta = 0$$

$$\Omega = \frac{I_1 \omega \pm \sqrt{I_1^2 \omega^2 - 4mgb(I_2 - I_1) \cos \theta}}{2(I_2 - I_1) \cos \theta}$$

[IF YOU USED THE DEFINITION $\omega_1 = \omega$, THEN $(I_2 - I_1) \rightarrow I_2$ ABOVE]

$$\text{FOR STEADY MOTION WE NEED } \omega > \frac{2}{I_1} \sqrt{mgb(I_2 - I_1) \cos \theta}$$

$$\omega > \frac{2}{a^2} \sqrt{gb(b^2 - \frac{a^2}{2}) \cos \theta}$$

$$\text{FOR LARGE } \omega, \Omega \rightarrow I_1 \omega \frac{1 \pm (1 - 2mgb(I_2 - I_1) \cos \theta) / I_1^2 \omega^2}{2(I_2 - I_1) \cos \theta} \quad \checkmark \quad \text{ROOT} = \frac{mgb}{I_1 \omega} = \frac{gb}{a^2 \omega}$$

I ACTUALLY HAD IN MIND A MORE DEVIOUS APPROACH:

IN THE ROTATING FRAME THE C.M. IS AT REST AND THE AXLE IS AT REST, ALTHOUGH THE HOOP SPINS ABOUT THE AXLE. THIS IS A KIND OF STATIC EQUILIBRIUM.

$$\therefore \sum \vec{F} = 0 \quad \text{AND} \quad \sum \vec{N} = 0 \quad \text{MUST HOLD IN THIS FRAME.}$$

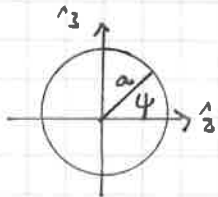
THE KEY IS $\sum \vec{N} = 0$. THERE ARE 3 KINDS OF TORQUES

$$\vec{N}_{\text{GRAVITY}} = -mgb \sin \theta \hat{z} \quad (\text{ABOUT THE PIVOT. WE USE THE SAME COORD SYSTEM AS ON P 6})$$

$$\vec{N}_{\text{CENTRIFUGAL}} = - \int dm \vec{r} \times (\vec{\Omega} \times (\vec{\Omega} \times \vec{r}))$$

$$\vec{N}_{\text{CORIOLIS}} = -2 \int dm \vec{r} \times (\vec{\Omega} \times \dot{\vec{r}}) = -2 \int dm \vec{r} \times (\vec{\Omega} \times (\vec{\omega} \times \vec{r}))$$

TO INTEGRATE AROUND THE HOOP, DEFINE ANGLE ψ IN THE \hat{z} - \hat{z} PLANE



$$\text{SO } dm = \frac{m}{2\pi} d\psi$$

$$\text{THEN } \vec{r} = (b, a \cos \psi, a \sin \psi)$$

$$\vec{\Omega} = (\Omega \cos \theta, \Omega \sin \theta, 0)$$

$$\vec{\omega} = (\omega, 0, 0)$$

$$\text{CENTRIFUGAL PIECE: } \vec{r} \times (\vec{\Omega} \times (\vec{\Omega} \times \vec{r})) = \vec{r} \times [(\vec{\Omega} \cdot \vec{r})\vec{\Omega} - \Omega^2 \vec{r}] = (\vec{\Omega} \cdot \vec{r})(\vec{r} \times \vec{\Omega})$$

$$\vec{\Omega} \cdot \vec{r} = b \Omega \cos \theta + a \Omega \sin \theta \cos \psi$$

$$\vec{r} \times \vec{\Omega} = \begin{vmatrix} \hat{1} & \hat{2} & \hat{3} \\ b & a \cos \psi & a \sin \psi \\ \Omega \cos \theta & \Omega \sin \theta & 0 \end{vmatrix} = -a \Omega \sin \theta \sin \psi \hat{1} + a \Omega \cos \theta \sin \psi \hat{2} + (b \Omega \sin \theta - a \Omega \cos \theta \cos \psi) \hat{3}$$

$$\text{SO } \vec{N}_{\text{CENT}} = -m \left\{ b^2 \Omega^2 \cos \theta \sin \theta - \frac{1}{2} a^2 \Omega^2 \sin \theta \cos \theta \right\} \hat{z}$$

$$\text{NOTING } \int d\psi \sin \psi = \int d\psi \cos \psi = \int d\psi \sin \psi \cos \psi = 0 \text{ ETC}$$

$$\text{CORIOLIS PIECE: } \vec{r} \times (\vec{\Omega} \times (\vec{\omega} \times \vec{r})) = \vec{r} \times [(\vec{\Omega} \cdot \vec{r})\vec{\omega} - (\vec{\Omega} \cdot \vec{\omega})\vec{r}] = (\vec{\Omega} \cdot \vec{r}) \vec{r} \times \vec{\omega}$$

$$\vec{r} \times \vec{\omega} = \begin{vmatrix} \hat{1} & \hat{2} & \hat{3} \\ b & a \cos \psi & a \sin \psi \\ \omega & 0 & 0 \end{vmatrix} = -a \omega \cos \psi \hat{3}$$

$$\text{SO } \vec{N}_{\text{CORIOLIS}} = + \frac{2M}{2} a^2 \omega \Omega \sin \theta \hat{z}$$

$$\text{SO } \sum \vec{N} = 0 = m(b^2 - \frac{a^2}{2}) \Omega^2 \cos \theta \sin \theta - ma^2 \omega \Omega \sin \theta + mgb \sin \theta$$

AS BEFORE....

(6) MY EQUATION (1'') HAD 2 ERRORS, WHICH SOME OF YOU NOTED BY COMMENT WITH SCHRÖDINGER'S (1'') AND (3)

$$(1'') \rightarrow f(\psi, \psi') = \left(\frac{\partial \psi}{\partial x}\right)^2 - \frac{2m}{\hbar^2} (E - V(x)) \psi^2$$

WE WANT $\delta I = \delta \int f(\psi, \psi') dx = 0$

THIS IS THE STANDARD EULER-LAGRANGE PROBLEM, WHOSE SOLUTION OBEYS LAGRANGE'S EQUATION:

$$\frac{d}{dx} \frac{\partial f}{\partial \psi'} - \frac{\partial f}{\partial \psi} = 0$$

$$\Rightarrow \left[\frac{\partial^2 \psi}{\partial x^2} + \frac{2m}{\hbar^2} (E - V) \psi = 0 \right]$$

IF YOU USED MY ORIGINAL FORM OF (1''), YOU WOULD GET

$$\frac{\partial^2 \psi}{\partial x^2} + \frac{m}{\hbar^2} (E + V) \psi = 0$$

WHICH IS NOT VERY SOLVABLE FOR V GIVEN AS BELOW

FOR $V = \begin{cases} 0 & 0 < x < a \\ \infty & \text{ELSEWHERE} \end{cases}$

WE SEE THAT OUTSIDE $(0, a)$, ψ MUST VANISH FOR FINITE E , SINCE $V \rightarrow \infty$ THERE.

ON $(0, a)$ WE HAVE $\psi'' + \frac{2mE}{\hbar^2} \psi = 0$

$$\Rightarrow \psi = A \cos \sqrt{\frac{2mE}{\hbar^2}} x + B \sin \sqrt{\frac{2mE}{\hbar^2}} x$$

TO BE CONTINUOUS AT $x=0$ AND a , ψ MUST VANISH THERE

$$\Rightarrow A=0, \quad \sqrt{\frac{2mE}{\hbar^2}} = \frac{n\pi}{a}$$

$$\text{OR } E = \frac{\hbar^2 k^2}{2m} \left(\frac{n\pi}{a}\right)^2$$

IF YOU BELIEVE $E = \frac{p^2}{2m}$ STILL HOLDS, THEN $p = \frac{n\pi \hbar}{a}$

AND $pa = n\pi \hbar$ SO $(pa)_{\text{MIN}} = \pi \hbar = \frac{h}{2}$

IF YOU FOLLOW SCHRÖDINGER: $p = \frac{\hbar}{i} \frac{\partial \psi}{\partial x} = \frac{n\pi \hbar}{a} \cos \sqrt{\frac{2mE}{\hbar^2}} x \approx \frac{n\pi \hbar}{a}$